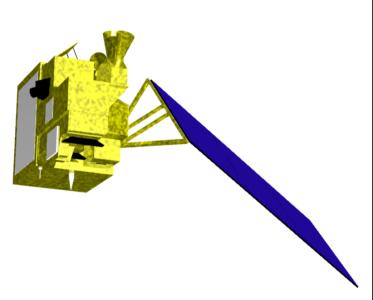
GOSAT Mission Objetives

- To contribute to policy development within Ministry of Environment (MOE) by estimating the source and sink of Green House Gases (GHGs) at subcontinental scale and by verifying the reduction of GHG's emission which is required by the Kyoto Protocol.
- To contribute to the advancement of earth observation technologies for future missions.
 - High reliability bus-system
 - High speed data handling system
 - High resolution Spectrometer
 (0.1cm⁻¹Over 20,000 channels)

GOSAT Targets

- Observation of CO2 density during the first commitment period (2008 to 2012) of the Kyoto Protocol for 3-month average with relative accuracy of 1% (4ppmv) at sub-continental spatial resolution.
- Reduction of errors by half in identifying the GHGs' source and sink at Sub-continental scale with the data obtained by GoSat in conjunction with that from the ground-based instruments.

GOSAT Spacecraft

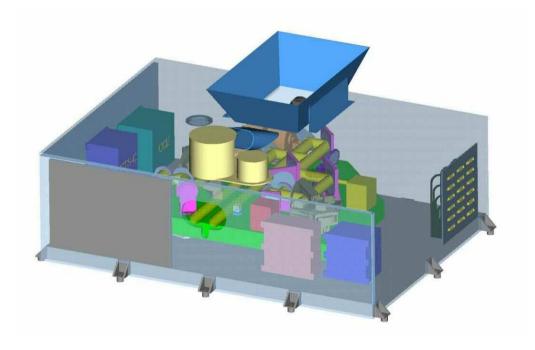


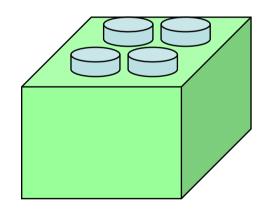
| size | 1.5m × 1.5m × 3.5m | | | | |
|------------------|-----------------------------|--|--|--|--|
| weight | 1650[kg] | | | | |
| Power | 300 W | | | | |
| Altitude control | 3 axes control | | | | |
| Telemetry and | (S band) Command :1[kbps] | | | | |
| Command | HK telemetry: 32.8[kbps] | | | | |
| | (X band) Mission Telemetry: | | | | |
| | 120 [Mbps] | | | | |
| life | 5 years | | | | |
| Orbit | Altitude 618 [km] | | | | |
| | Inclination 65 deg | | | | |
| | Non Sun Synchronous | | | | |
| | (3 days revisit) | | | | |
| Launch | Feb, 2008, by H-II rocket | | | | |

Mission Sensor

Greenhouse gases Observing Sensor

Cloud and Aerosol Sensor





Specifications of Greenhouse gases Observing Sensor

| Ground F | ointing | Configu | ıration | 2-axes scanner (fully redundant) | | | | | |
|--|------------------------------|---------------------------|---------------------------|---|-----------|-----------|-----------------------|----------|--|
| Mechanis Fore option | | Scanning Field of view | | Cross Track (±35 deg) Along Track (±20deg) | | | | | |
| | | | | IFOV 8 km、88 km (Interval) 790 km (scan width) (latitude of 30 deg) | | | | | |
| Fourier Transfo rm Spectro meter | Speed | | 0.7~1 (Interferogram)/sec | | | | | | |
| | Spectral band | | 1 | | 2 | 3 | 4 | 5 | |
| | Coverage (cm ⁻¹) | | 12900- 13200 | | 5200-6400 | 4800-5200 | 2000-2500 | 660-2000 | |
| | resolutio | n (cm ⁻¹) 0.5 | | | 0.2 | 0.2 | 0.1 | 0.1 | |
| | SNR | | | | ~600 | | | | |
| | Detector | Detector | | | InGaAs | InGaAs | InSb | PC-MCT | |
| | Calibration | on | Solar Iri | adiance, Deep Space, Moon | | | Blackbody, Deep space | | |

Detector: 2 per band for the redundancy and polarization

Spectral Coverage

GOSAT Spectral Coverage

